

Model Documentation of the 'ASTOVL Aircraft'

1 Nomenclature

1.1 Nomenclature for Model Equations

- x state vector
- u control input vector
- w noise vector
- z regulated output vector
- y measurement vector

2 Model Equations

State Vector and Input Vector:

$$x \in \mathbb{R}^{28} \quad u \in \mathbb{R}^3 \quad w \in \mathbb{R}^{28} \quad z \in \mathbb{R}^{28} \quad y \in \mathbb{R}^4$$

System Equations:

$$\dot{x}(t) = Ax(t) + B_1w(t) + Bu(t) \tag{1a}$$

$$z(t) = C_1x(t) + D_{11}w(t) + D_{12}u(t) \tag{1b}$$

$$y(t) = Cx(t) + D_{21}w(t) \tag{1c}$$

Outputs: z

3 Derivation and Explanation

This model is part of the "COMpleib" - library and was automatically imported into ACKREP.

The original description was:

AC13 ASTOVL Aircraft, LARGE model save ASTOVL_aug A B1 B2 C1 C2 D11 D12 D21 D22 Ag Bg Cg Dg in ASTOVL.m /export/home/leibfr/Lipinski/matlab

4 Simulation

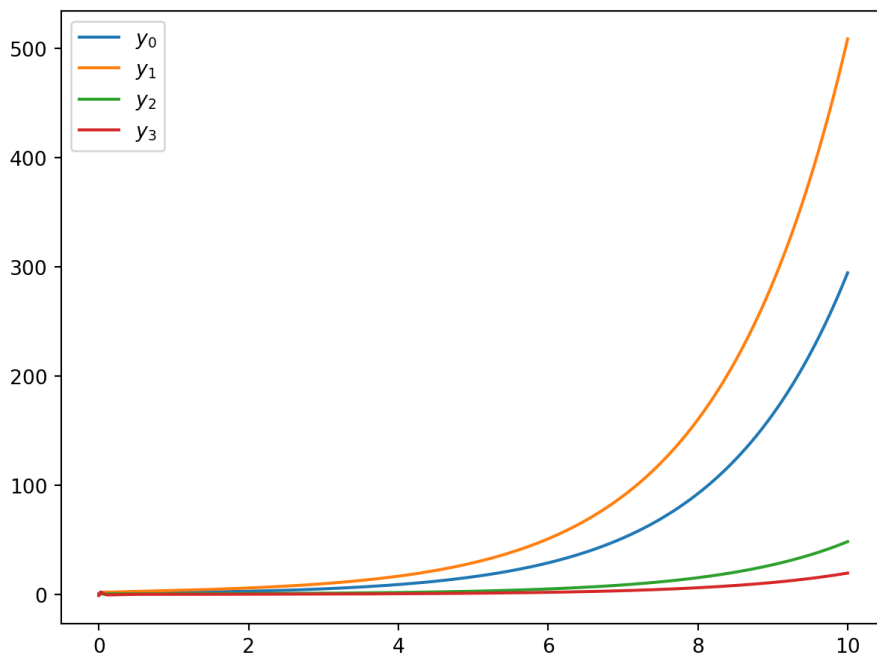


Figure 1: Simulation of the ASTOVL Aircraft.

References

- [1] . Toffner-Clausen, "System Identification and Robust Control A Case Study Approach", Springer-Verlag, "Advances in Industrial Control", 1996 p. 274