

Model Documentation of the 'Aero engine control'

1 Nomenclature

1.1 Nomenclature for Model Equations

- x state vector
- u control input vector
- w noise vector
- z regulated output vector
- y measurement vector

2 Model Equations

State Vector and Input Vector:

$$x \in \mathbb{R}^2 u \in \mathbb{R}^3 w \in \mathbb{R}^6 z \in \mathbb{R}^9 y \in \mathbb{R}^6$$

System Equations:

$$\dot{x}(t) = Ax(t) + B_1w(t) + Bu(t) \quad (1a)$$

$$z(t) = C_1x(t) + D_{11}w(t) + D_{12}u(t) \quad (1b)$$

$$y(t) = Cx(t) + D_{21}w(t) \quad (1c)$$

Outputs: z

2.1 Exemplary parameter values

Parameters omitted due to large matrices. See Source code.

3 Derivation and Explanation

This model is part of the "COMPleib" - library and was automatically imported into ACKREP.

The original description was:

JE3 Aero engine control "Multivariable feedback control Analysis and design"
S. Skogestad and I. Postlethwaite John Wiley and Sons, 1996, Section 12.3.3
Note Matlab files <http://www.nt.ntnu.no/users/skoge/book/matlab.html> stored
in /export/home/leibfr/Lipinski/matlab/... Examples_Multi_Feedback_Control/matlab.m/
F. Leibfritz, 16.09.2003 Data matrices generated by Sec12_33.m in directory
above on Laptop save Aero_Engine a b c d A_Hinf B1 B2 C1 C2 D11 D12 D21
D22 A_Hinf B1 B2 C1 C2 D11 D12 D21 D22 – data set G5 for H_inf design
==> A_Hinf = A in Sec12_33.m

4 Simulation

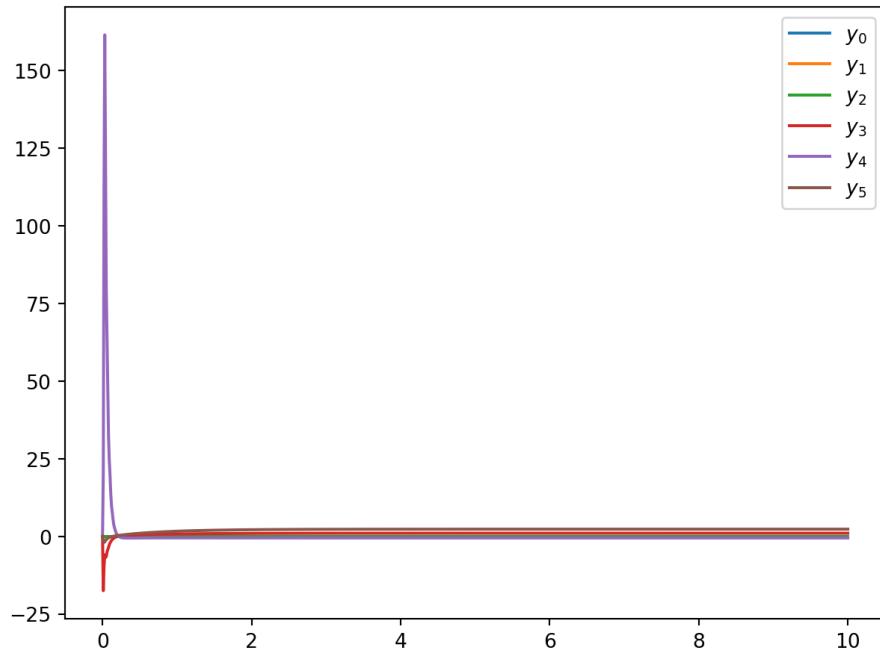


Figure 1: Simulation of the Aero engine control.

References

- [1] Multivariable feedback control Analysis and design" S. Skogestad and I. Postlethwaite John Wiley and Sons, 1996, Section 12.3.3